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Wind Tunnel Technique. Pankhurst and Holder . Pitman, London, 1952. 702 pp. 373 figures. 57s. 6d. net. - Volume 57 Issue 505 - J. H. Preston

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## **Innovative approach to design a new national low speed ...**

Pankhurst R. C. and D. W. H., Wind-Tunnel Technique: An Account of Experimental Methods in Low- and High-Speed Wind Tunnels, 1st ed., Sir Isaac Pitman and Sons, Pitman, London, 1952, Chap. 3.

## **Wind Tunnel Investigation of a Circulation Control Wing ...**

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Pope and Harper - later, less comprehensive; a still later edition may exist with a slightly different authorship. For really slow wind tunnels, there's an appendix in the 1st edition of my own "Life in Moving Fluids" - 1981. (Not in 2nd ed.) Steven Vogel  
Undisciplined ...

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Most widely held works about R. C Pankhurst The measurement of air flow by E Ower ... Wind-tunnel technique; an account of experimental methods in low- and high-speed wind tunnels by R. C Pankhurst ( Book ) 20 editions published between 1952 and 1968 in English and Italian and held by 292 WorldCat member libraries worldwide Dimensional analysis and scale factors by R. C Pankhurst ( Book ) 18 ...

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Pankhurst, R. C. et al., Wind-Tunnel Technique, London, Pitman & Sons, Ltd., 1952, pp. 140-148.

## **Smoke generator - PARRISH; KEMP L.**

Comments, mostly cautionary, are made on the features of certain existing tunnels, and drawings and tabulated data are given for about thirty typical tunnels of various ages. The emphasis is on low-speed tunnels of wooden construction, but much of the discussion is applicable to the design of tunnels of larger size or higher speed.

## **The design of low-speed wind tunnels - ScienceDirect**

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### **Performance Measurements of an Airfoil at Low Reynolds ...**

Mehta, R D and Bradshaw P (1979) Technical notes: Design rules for small low speed wind tunnels, the Aeronautical Journal of the Royal Aeronautical Society, vol. 83 pp442-449

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Wind tunnel tests were conducted to determine the low speed two dimensional aerodynamic characteristics of a 17 percent thick medium speed airfoil (MS(1)-0317) designed for general aviation applications. The results were compared with data for the 17 percent thick low speed airfoil (LS(1)-0417) and the 13 percent thick medium speed airfoil (MS(1)-0313). Theoretical predictions of the drag rise characteristics of this airfoil are also provided. The tests were conducted in the Langley low turbulence pressure tunnel over a Mach number range from 0.10 to 0.32, a chord Reynolds number range from 2 million to 12 million, and an angle of attack range from about -8 to 20 deg.

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